

Alternative Bio-Derived JP-8 Class Fuel and JP-8 Fuel: Flame Tube Combustor Test Results Compared using a GE TAPS Injector Configuration

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Motivation and Objectives



High oil costs + the need to reduce pollution and dependence on foreign suppliers has spurred great interest and activity in developing alternative aviation fuels

NASA Fundamental Aeronautics supported efforts in studying the effects of fuel alternatives in combustion and in engines, including

Alternative-Fuel Effects on Contrails and Cruise Emissions (ACCESS)

Alternative Aviation Fuels Experiment (AAFEX)

NASA ERA supports alternative fuels research: develop and demonstrate a low NO_x *Fuel flexible combustor* that provides a 75% reduction in oxides of nitrogen below the current CAEP 6 standard with no increase in particulate matter, while achieving a 50 percent reduction in fuel burned

Task objectives

—using GE TAPS single cup flame tube as a test bed

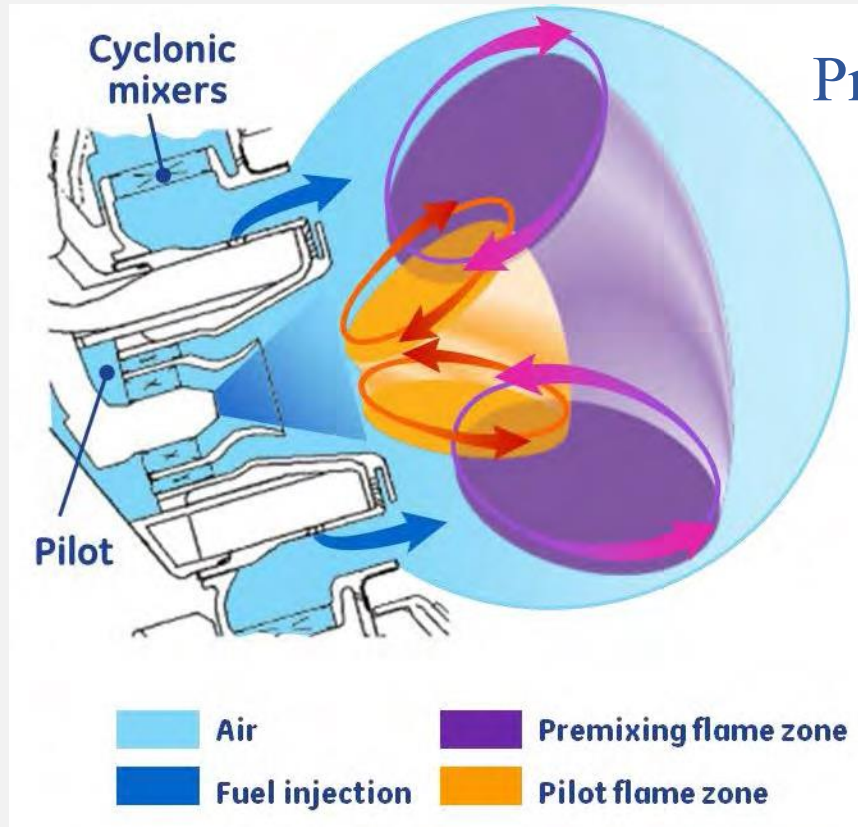
Ascertain visible luminosity, sooting, fuel spray pattern, liquid fuel penetration, flame zone location of Hydrotreated Renewable Jet (HRJ) fuel compared to JP-8.

Means: 1. high-speed imaging (grey scale) for structure, flame length, luminosity

2. Planar laser scatter of fuel drops

3. Fuel and OH planar laser-induced fluorescence (PLIF)

GE Twin Annular Premixing Swirler (TAPS) injector concept for low NO_x emissions



Provides independent control of:

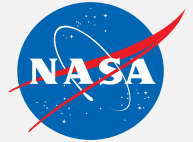
- Center pilot for low power operability, low CO, HC emissions
- Cyclone/main for high power operation, low NO_x emissions

References:

Foust, Thomsen, Stickles, Cooper, Dodds—AIAA 2012-0936

Mongia—AIAA 2003-2657

Comparing fuel physical properties



components

Fuel	JP-8	HRJ
Sulfur (ppm)	1148	<3
Olefins (%vol)	0.9	0.4
Aromatics (%vol)	18.6	0.4
Naphthalenes (%vol)	1.6	0
Initial boiling point, °	158	165
10%	176	179
90%	248	243
End Point	273	231
Flash Point °C	46	55
API Gravity	41.9	54
Specific Gravity	0.816	0.758
Freezing Point °C	-50	-62
Viscosity	4.7	5.3
Cetane Index	41	67
H Content (%mass)	13.6	15.3
Heat combustion (MJ/kg)	43.3	44.5
Fuel H/C ratio	1.88	2.12

Similar reactivity

HRJ:

Fewer aromatics:

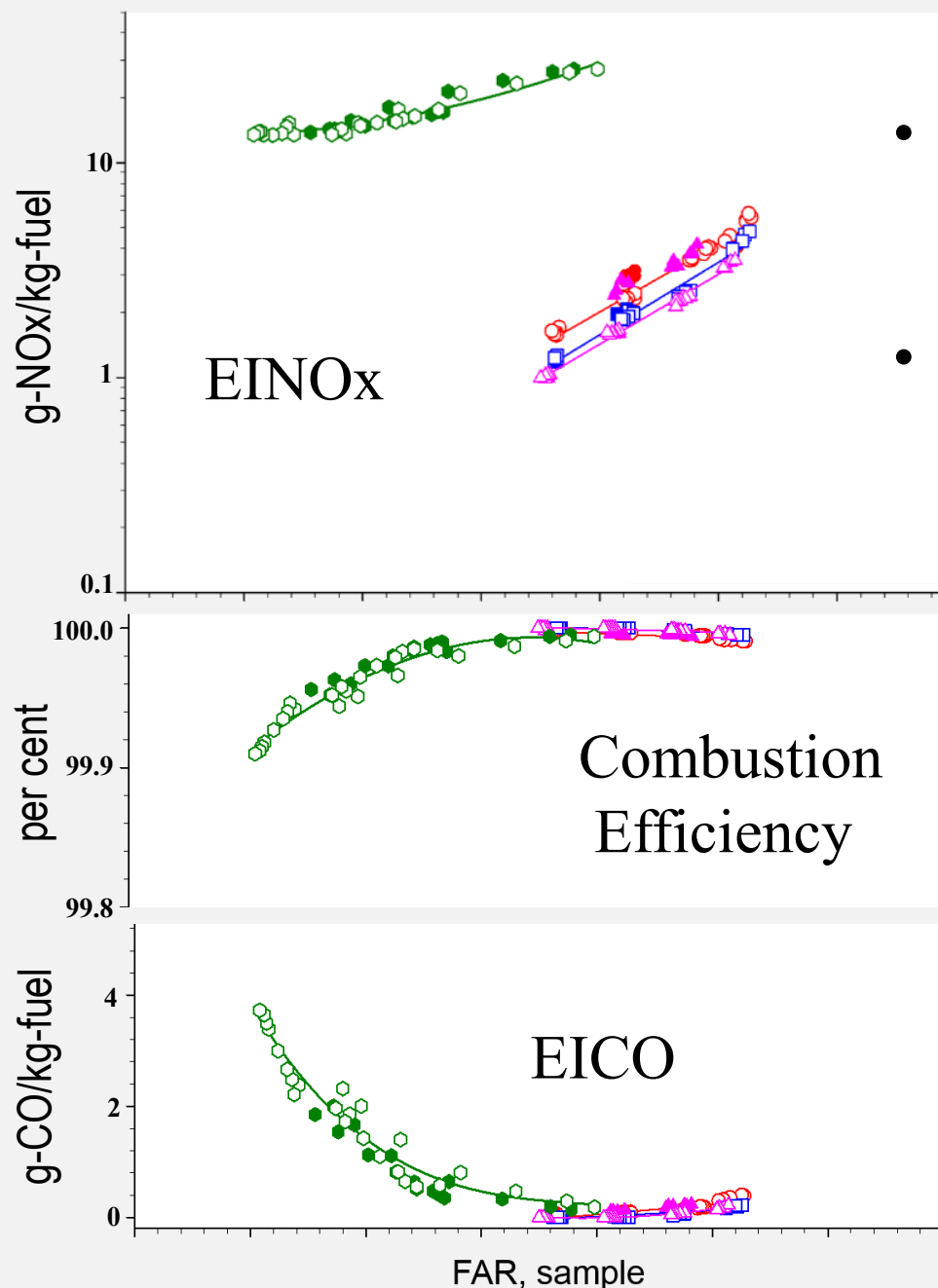
Less luminous

Higher cetane index:

Shorter ignition delay

Expectation: more **soot** production from **JP-8**—greater luminosity
HRJ constituents—shorter ignition delay time

TAPS Gaseous Emissions Results—Fuel type comparison



- No discernable difference between fuel types in combustion efficiency or emissions.
- This result is similar to other fuel comparison tests using different fuel-air mixers.

Legend: P₃, T₃, pilot/main split, fuel

- 250 psia, 1000°F, 10/90 split, HRJ
- 250 psia, 1000°F, 10/90 split, JP8
- 250 psia, 1000°F, 10/90 split, HRJ
- 250 psia, 1000°F, 10/90 split, JP8
- ▲ 170 psia, 1000°F, 10/90 split, HRJ
- △ 170 psia, 1000°F, 10/90 split, JP8
- ◆ 208 psia, 1000°F, 100/0 split, HRJ
- ◇ 208 psia, 1000°F, 100/0 split, JP8

A schematic diagram of the experimental setup for laser velocimetry. The setup includes a laser sheet (indicated by a dashed line) passing through a test section. An air flow (indicated by an arrow) enters the test section from the left. The test section is equipped with fused-silica windows and a filter wheel. Two ICCD cameras are used for imaging: one labeled 'CD camera' and another labeled 'ICCD camera' with the model number 'CD-10-83238'. The coordinate system is defined with +x, +y, and +z axes.

Diagram illustrating the flow and mixing region (FOV) between the Main burner and the Pilot burner. The Main burner is shown at the top, and the Pilot burner is shown at the bottom. The FOV (Flow of Vorticity) is indicated by a black box and labeled as the mixing region between the pilot and main burners.

- Planar Laser induced fluorescence (PLIF) of OH and Fuel, 100-ns gate
- Planar Laser Scatter (PLS) for Liquid Fuel, 100-ns gate
- Instantaneous imaging of CH* chemiluminescence, 100-ns, 100- μ s
Camera: Princeton Instruments PIMAX, 1k x 1k pixel

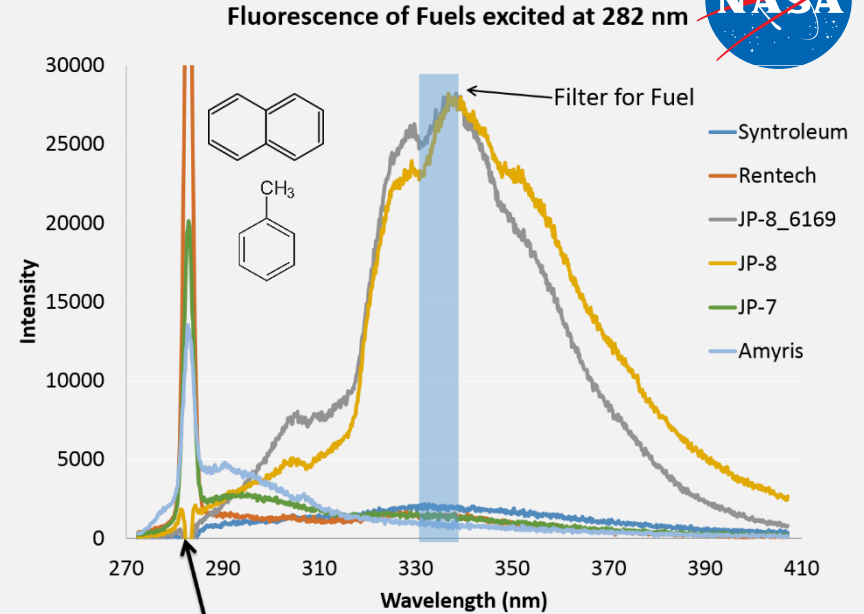
- High Speed Flame Imaging via Chemiluminescence of CH^* , C_2^*
Camera: Photron Fastcam SA1, 1k x1k px, 10000 frames/s

Optical diagnostics expectations based on fuel composition



Fuel	JP-8	HRJ
Sulfur (ppm)	1148	<3
Olefins (% vol)	0.9	0.4
Aromatics (% vol)	18.6	0.4
Naphthalenes (% vol)	1.6	0

- more aromatic content in JP-8:
higher **soot** production
→ **greater luminosity**

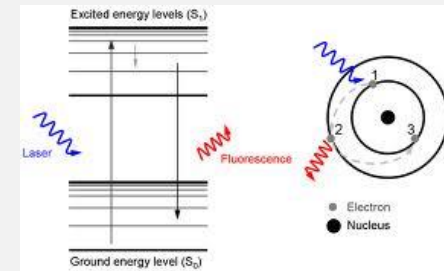


Laser Wavelength Used for OH, fuel excitation)

Light is either absorbed, scattered, or transmitted through matter.

$$I_{\text{trans}} = I_{\text{incident}} - I_{\text{absorbed}} - I_{\text{scattered}}$$

PLIF requires **absorption** by fuel constituents before the excited molecules can emit light

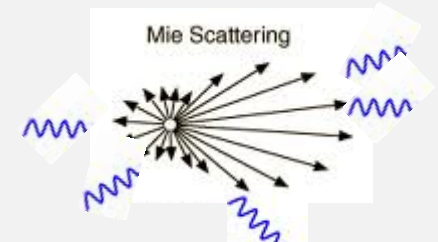


Naphthalenes and Methylbenzenes used for Fuel PLIF, so

- fuel PLIF signal greatly reduced for HRJ
- OH PLIF signal may be increased

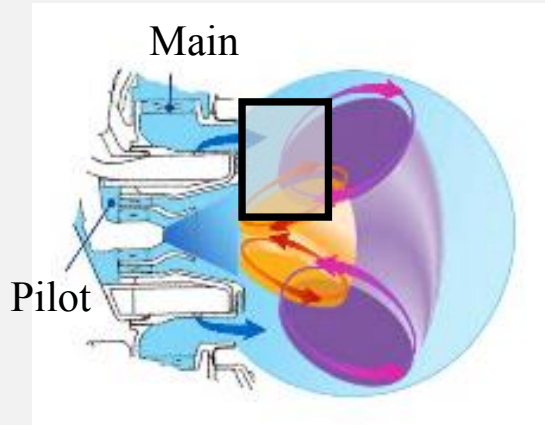
but More laser energy available for scattering from liquid

- PLS signal increased



Flame Chemiluminescence:

CH^*



% pilot flow affects
flame structure

Point 1
100/0
650°F

Point 2
10/90
925°F

Point 3
20/80
1000°F

Point 4
10/90
1000°F

Distance from center

Distance from center

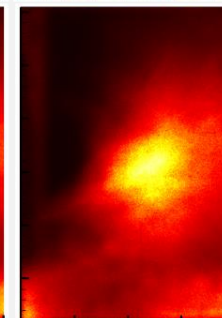
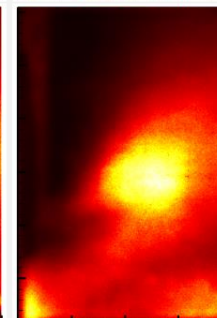
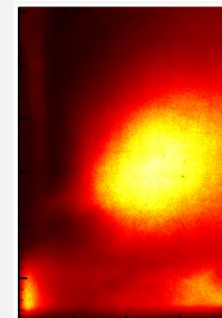
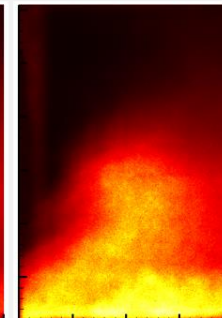
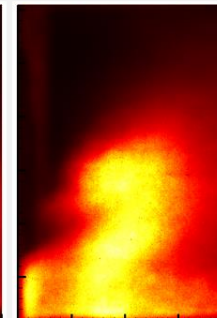
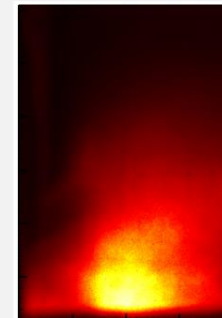
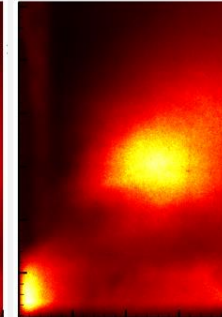
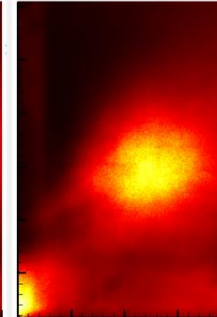
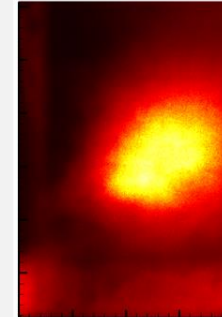
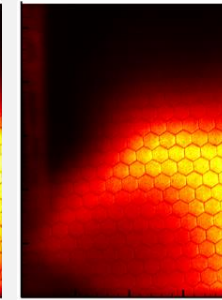
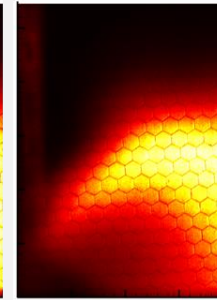
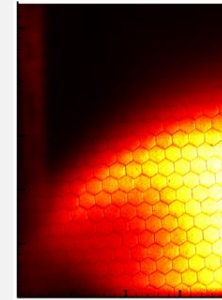
Distance from center

Distance from center

JP8

75% HRJ

HRJ



CH^*

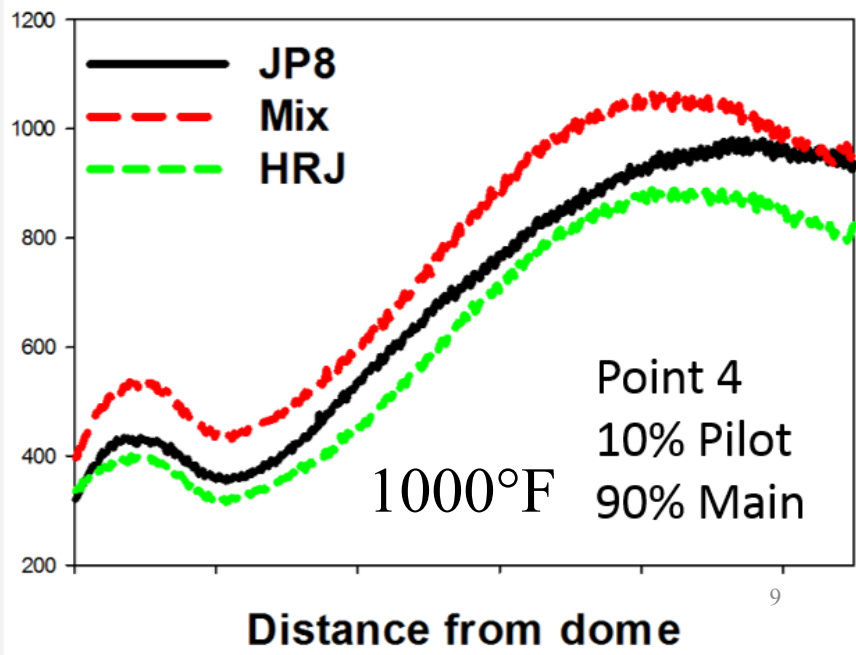
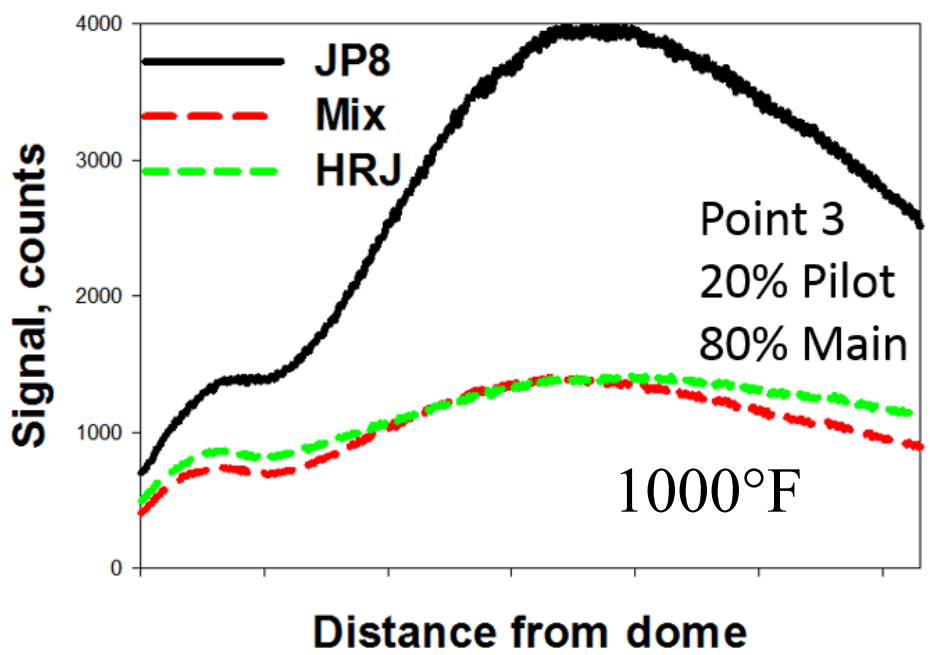
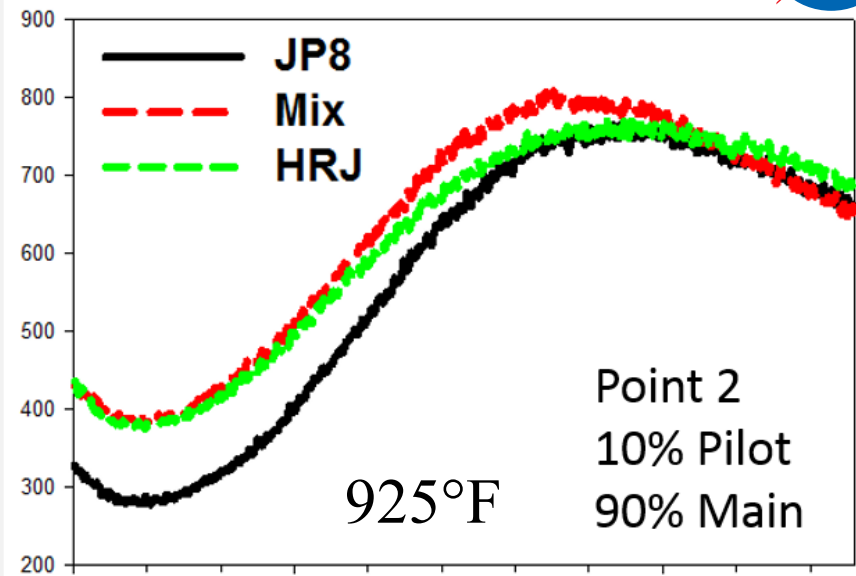
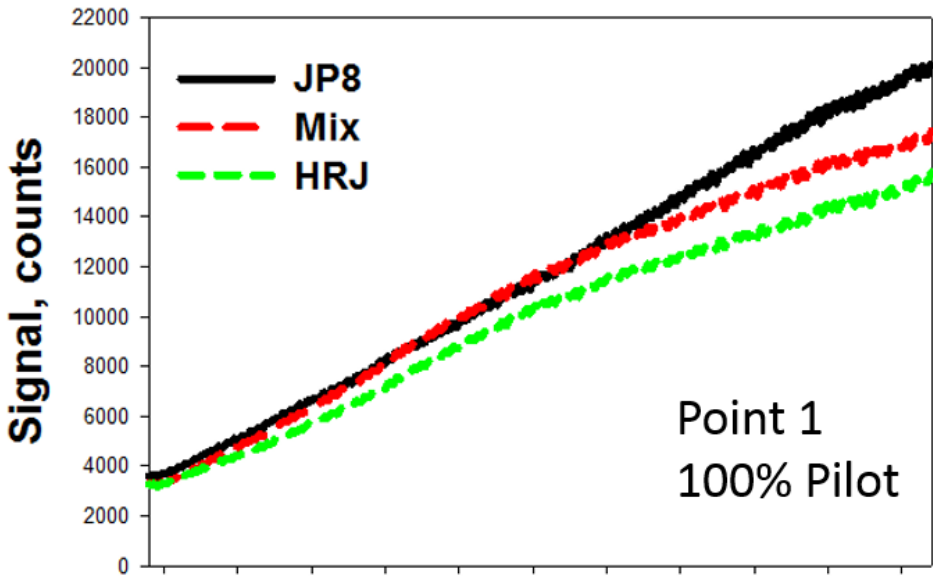
high

low

Distance from dome →



CH* results—total signal per image column with downstream location



High speed flame imaging— C_2^ , CH^* pilot only*



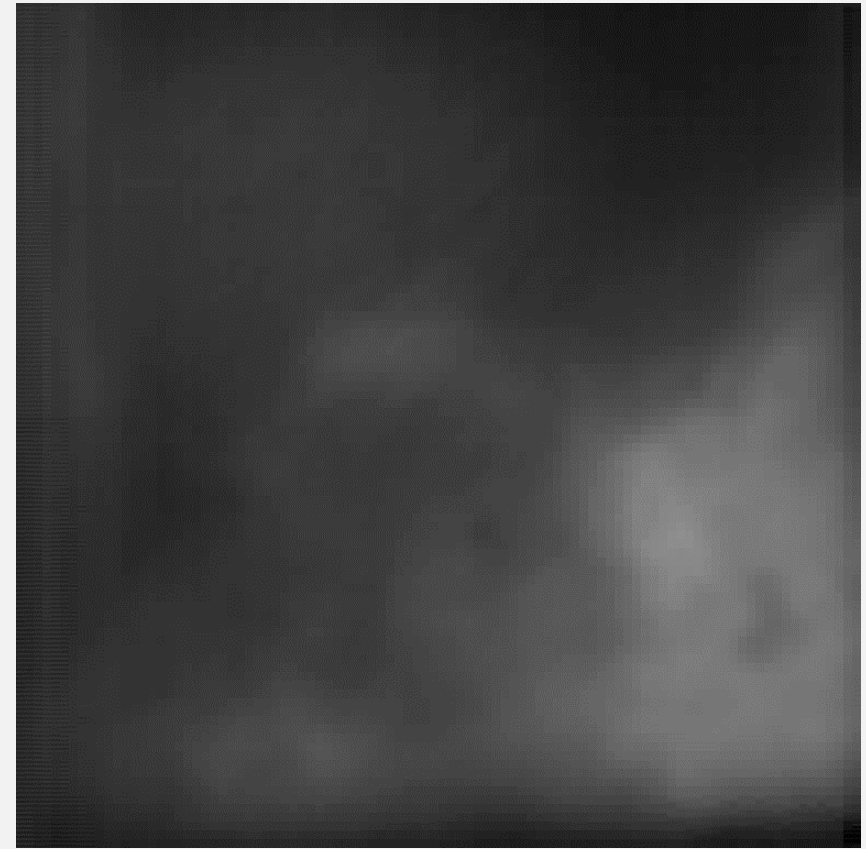
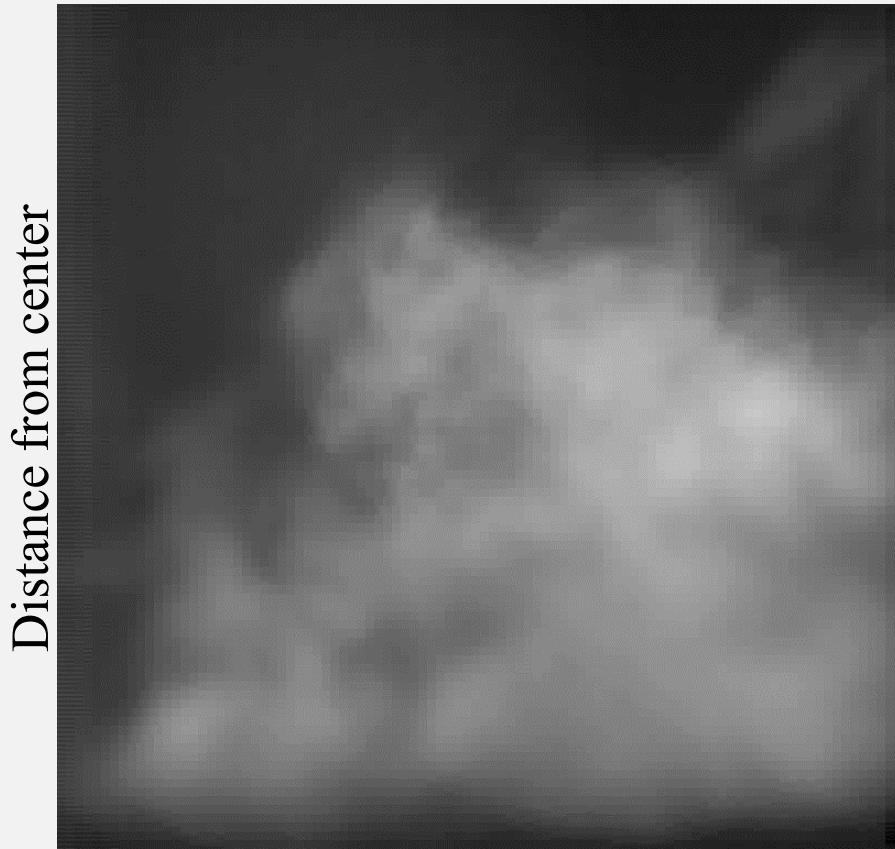
Frame rate: 10000/sec, 100- μ s exposure

Image Resolution 768 x 768 pixels

Flow direction left to right

JP-8

HRJ



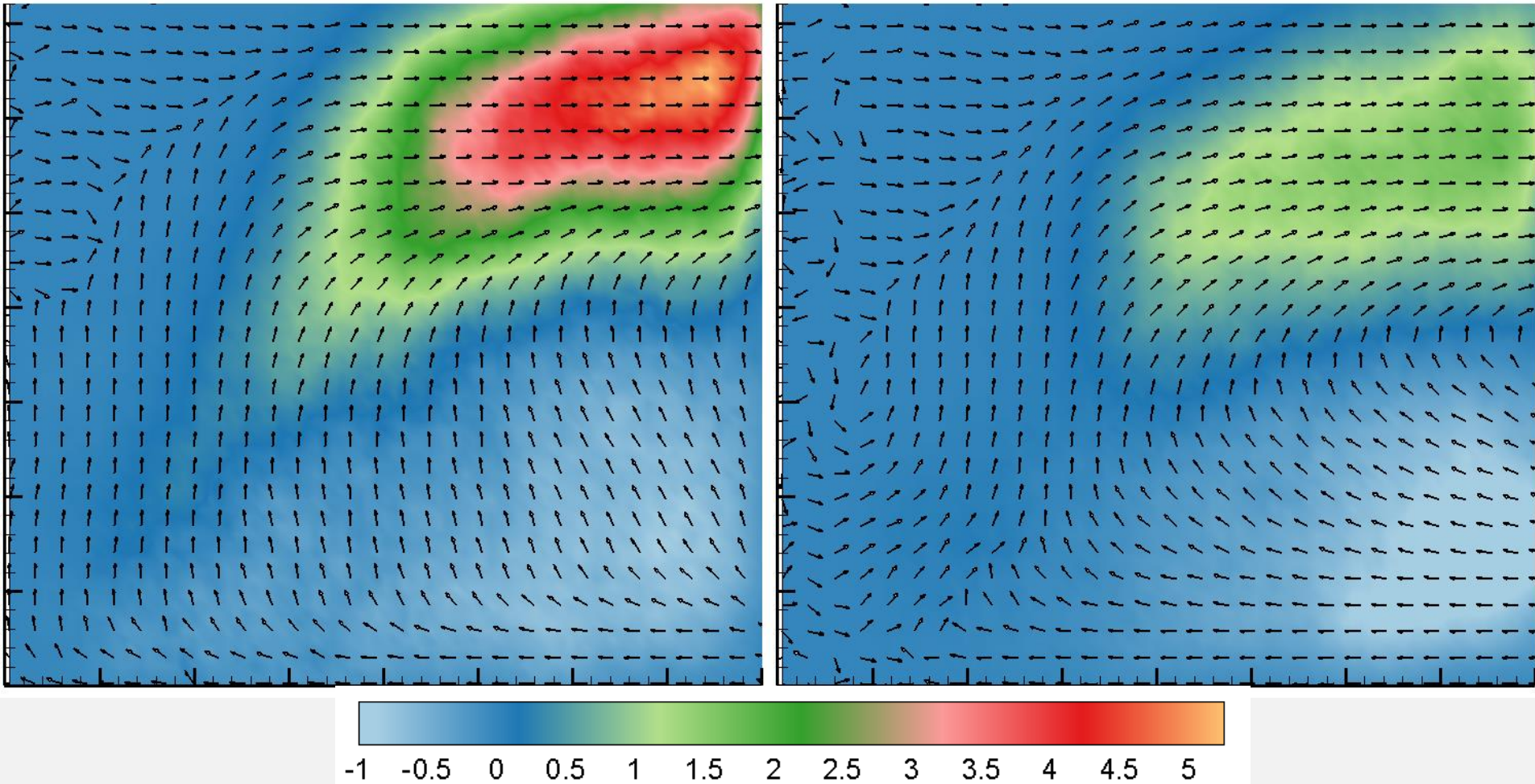
- JP-8 flame more luminous than HRJ flame
- Central recirculation zone can be seen

High speed video results—100% pilot, test point 1

- high speed camera frames (9701 images) processed as time-resolved PIV
- flow: left to right

JP-8

HRJ



Vectors give bulk average direction of motion—correspond visually
Contour shows the relative degree of change, on average

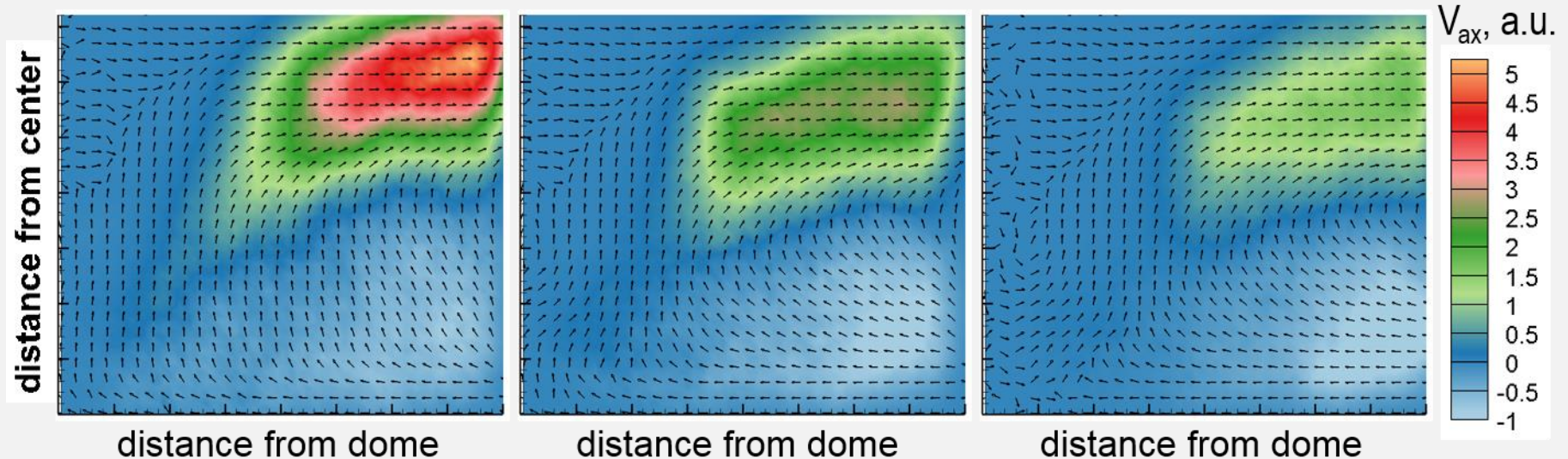
High speed video results—100% pilot, test point 1

Fuel Mixture shows results intermediate to the neat fuels

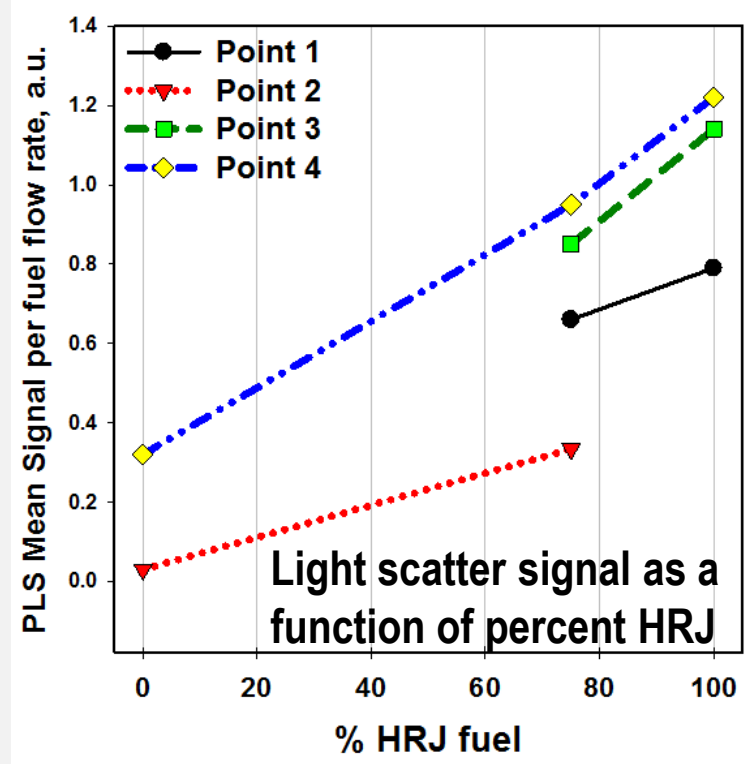
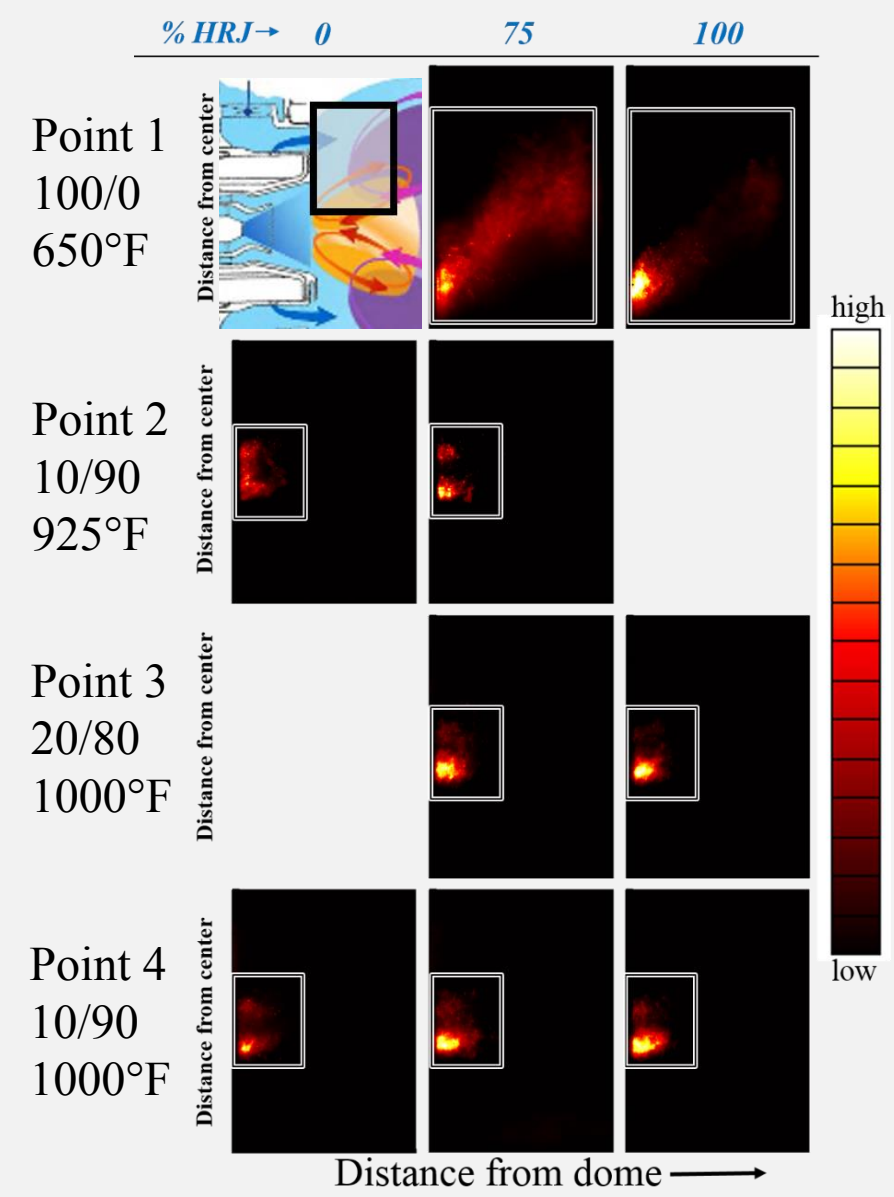
JP-8

75%-HRJ

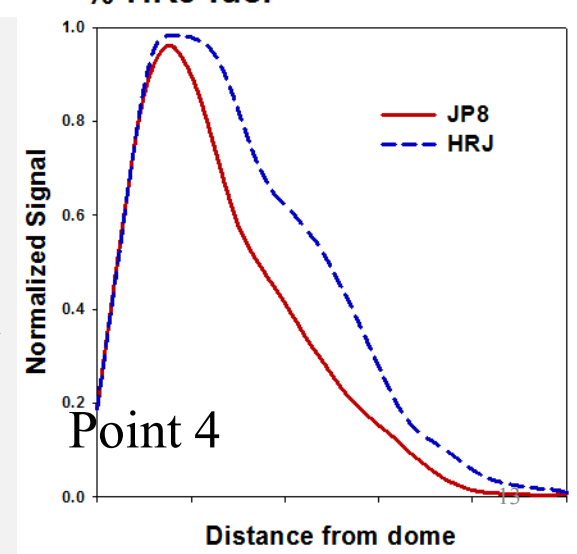
HRJ



Liquid fuel results, Planar Laser Scatter

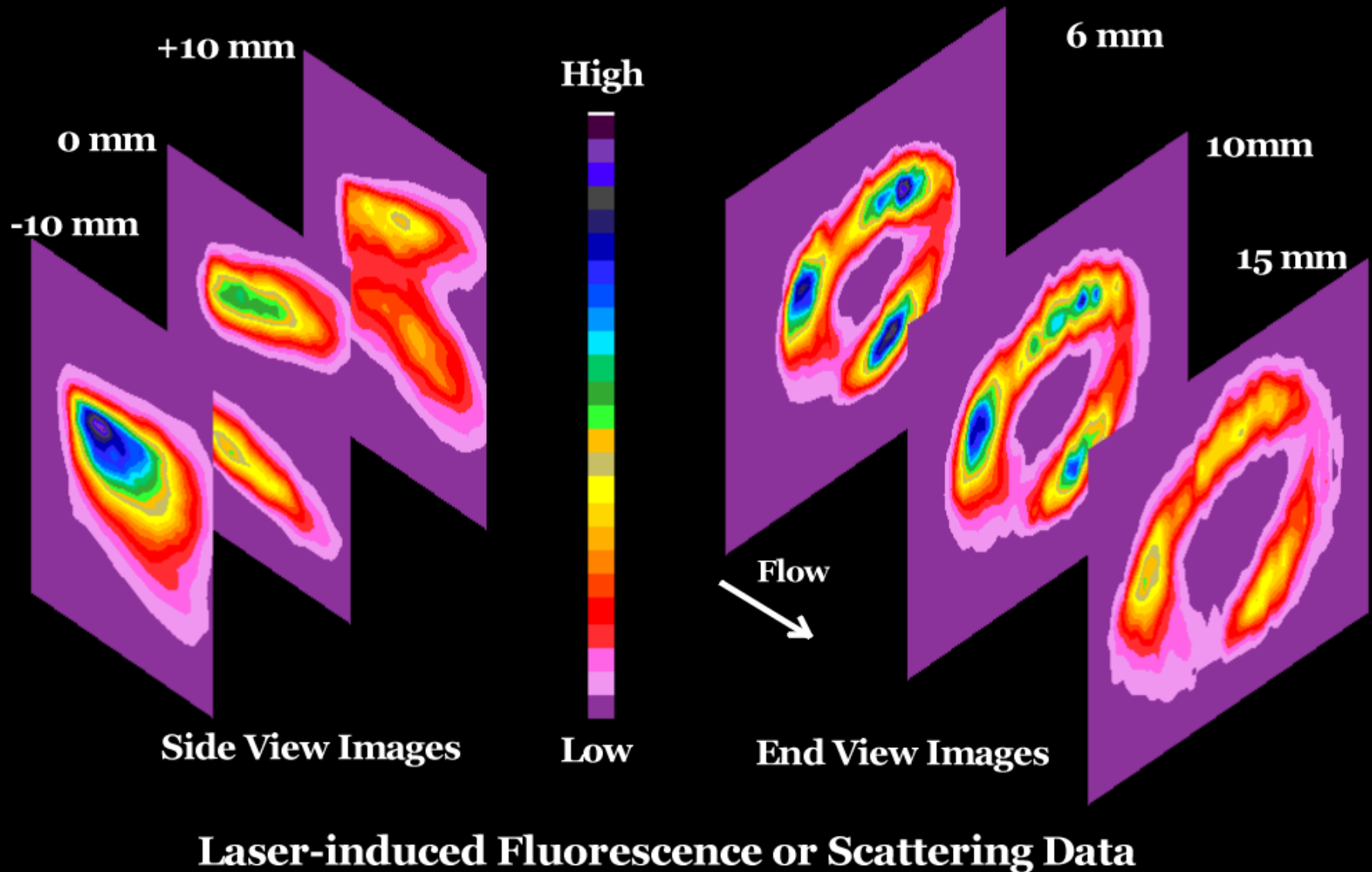


Liquid signal decay →



Above: average of 100 single shot images

Next: OH, fuel PLIF Results and Field of View Perspective:



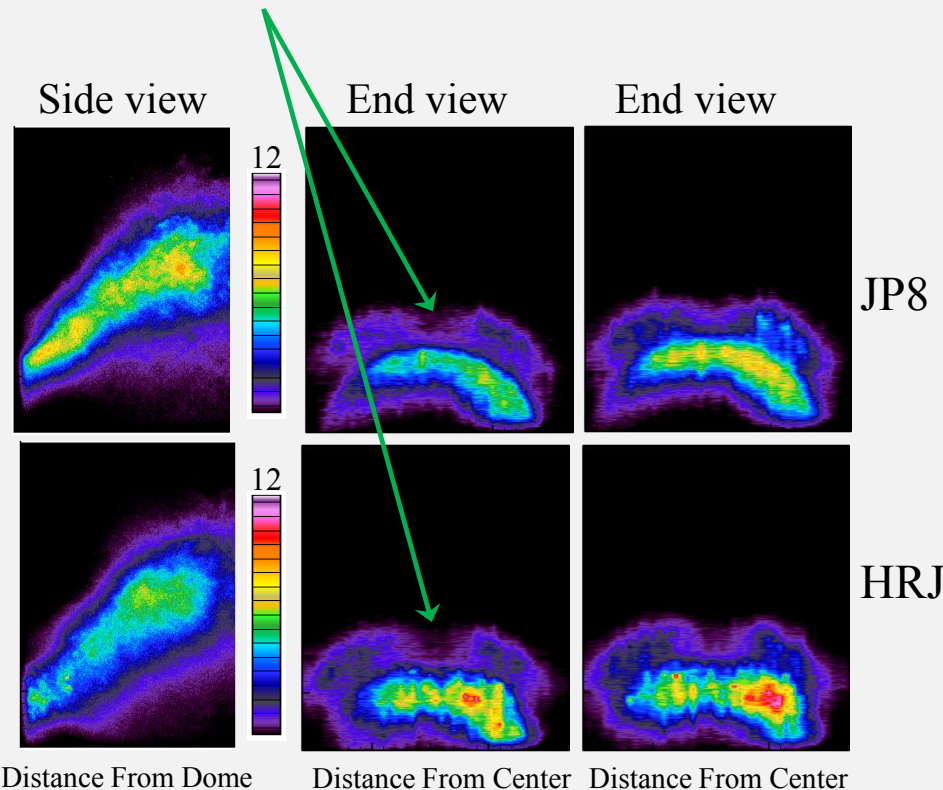
Left: laser sheet oriented with flow, traversed across flow, side view images

Right: resulting traverse block sliced at fixed axial positions to produce End View images

Fuel PLIF

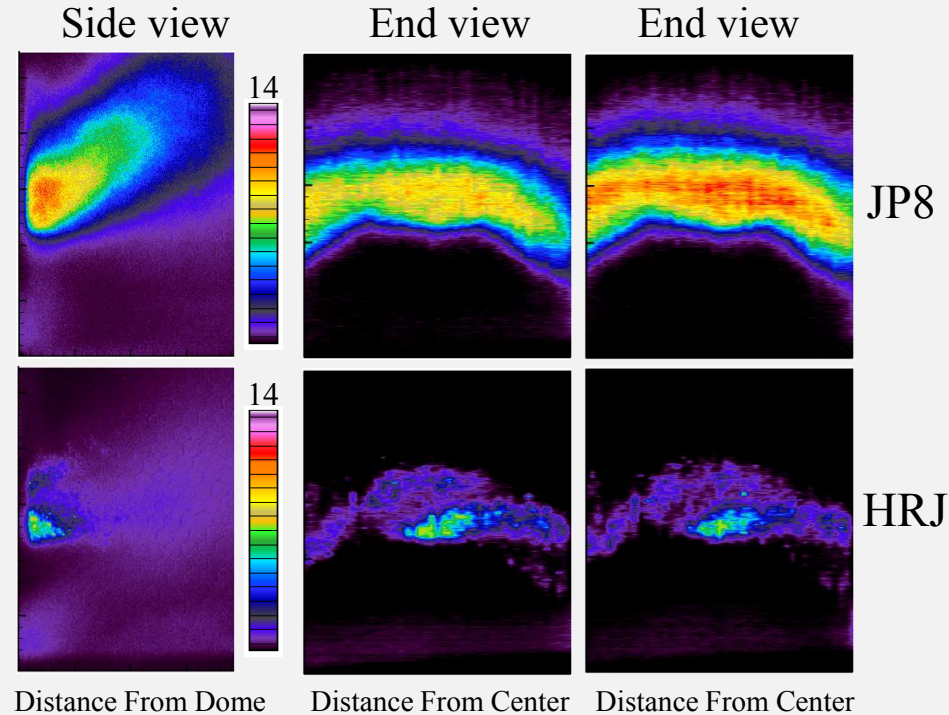
Point 1, pilot only

Notch due to N2 purge



For pilot only, JP8, HRJ have similar spray pattern

Point 2, 10/90 split, $T_{in} = 925^{\circ}\text{F}$



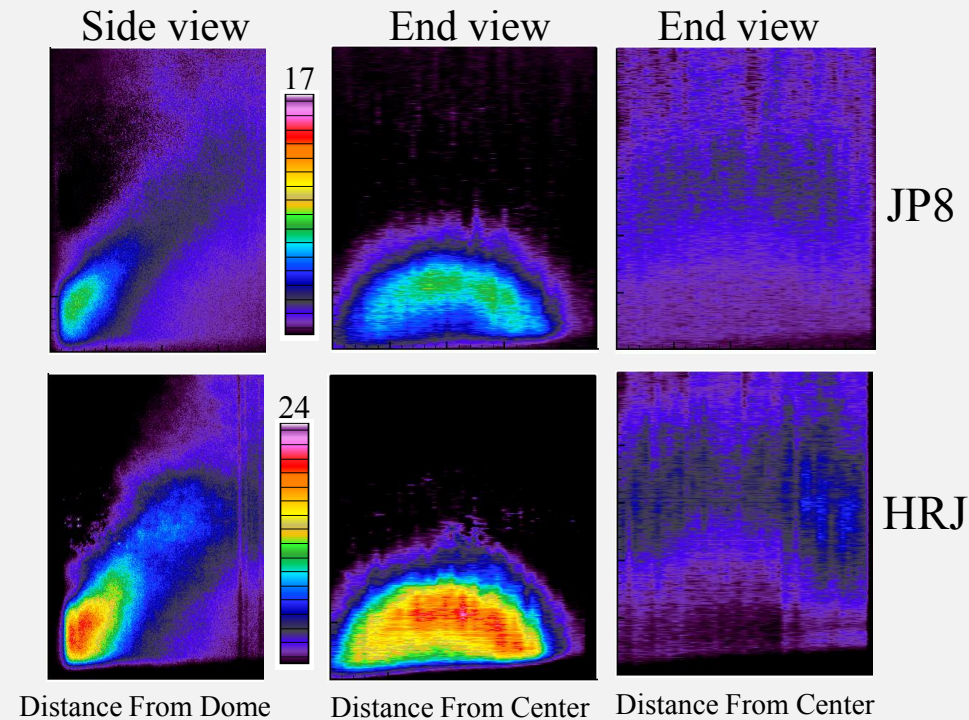
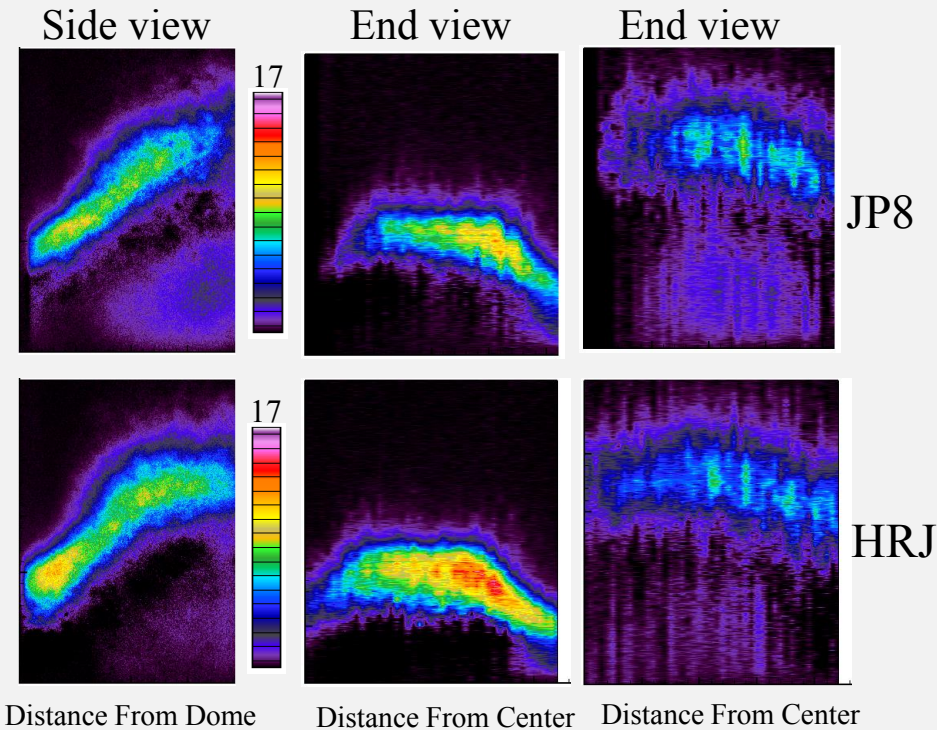
JP8: uniform distribution within annulus
HRJ: Most fuel observed near wetted annular walls

Possibly only HRJ liquid seen because greater number density than in gas phase

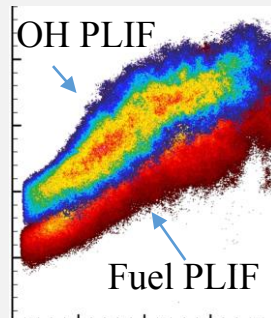
OH PLIF

Point 1, pilot only

Point 2, 10/90 split, $T_{in} = 925^{\circ}\text{F}$



OH PLIF signal not as strong in CRZ for HRJ, but stronger on air side of spray cone:

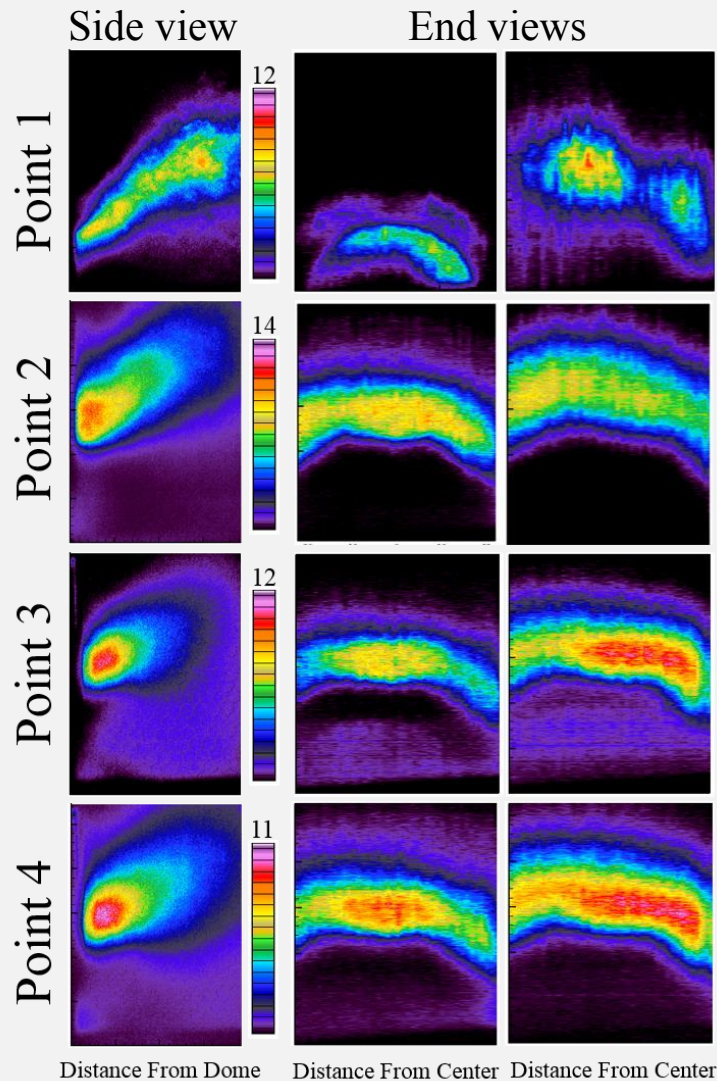


- Similar patterns for JP8, HRJ
- HRJ signal stronger than for JP8 likely because little absorption by fuel

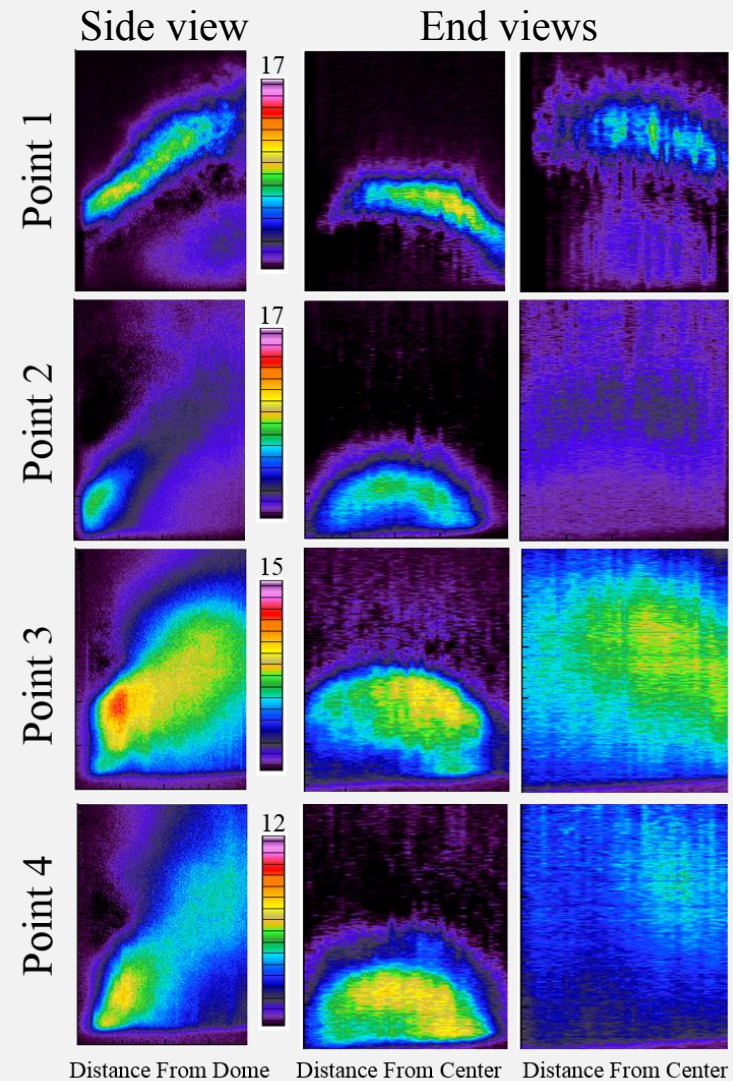
JP8: Compare four test points using PLIF



Fuel PLIF



OH PLIF





Summary

- A single-cup GE TAPS injector used to compare JP-8 and tallow HRJ, using sample gas analysis, flame chemiluminescence, PLS, OH and fuel PLIF
- Consistent with other flame tube combustor and engine tests, little or no difference in gaseous emissions of NO_x, CO, UHC
- Flame luminescence shows flame structure changes most affected by pilot flow. JP-8 flame ~4x brighter than HRJ flame for 20/80 split. Other splits have comparable luminosity
- When flow is split between pilot and main, we see liquid from main circuit but not from the pilot. Main circuit fuel does not completely vaporize before exiting the dome.

Summary, cont



Fuel PLIF:

- For HRJ, more fuel observed along the wetted walls of annulus, whereas with JP8, uniformly distributed
- Likely for HRJ, PLIF results primarily from the liquid phase, where the number density of aromatics is greater than in the gas phase
- Future Fuel PLIF with low aromatic fuel will need to use shorter wavelengths (~ 266 nm) for better signal

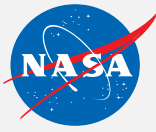
OH PLIF:

- Similar patterns are observed for both fuels in the flow split case. Under pilot only operation, little OH is observed in the central recirculation zone for the HRJ



Acknowledgment

- The NASA ARMD Environmentally Responsible Aviation Program sponsored this work.
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Questions?